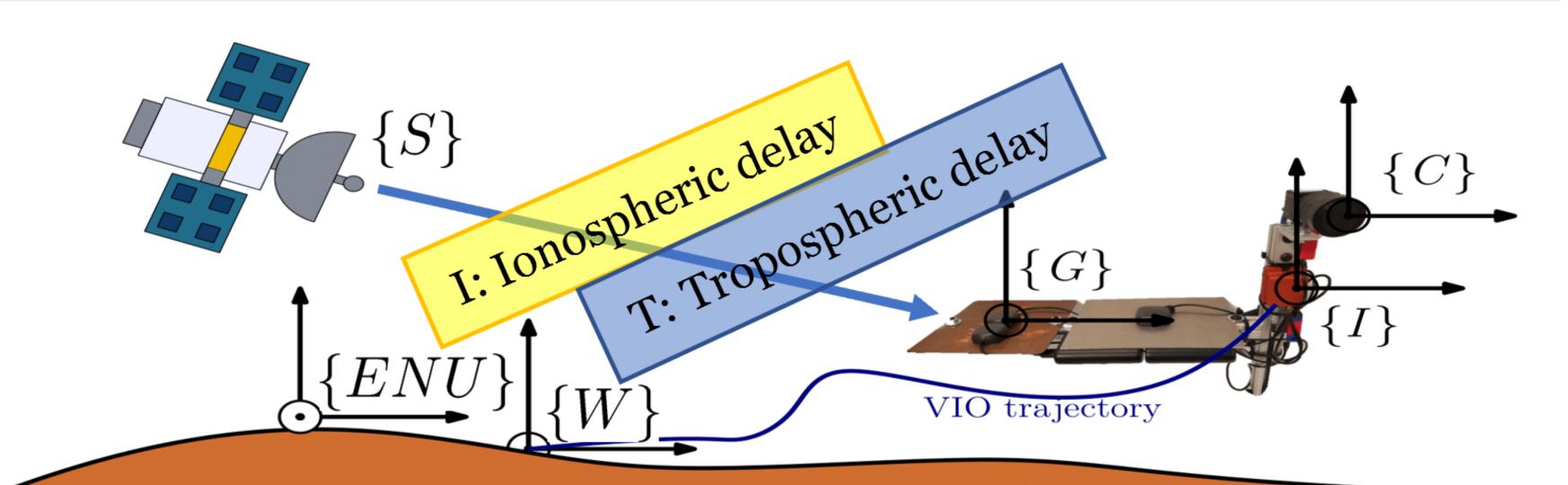
# **Tightly-coupled GNSS-aided Visual-Inertial Localization** Woosik Lee, Patrick Geneva, Yulin Yang, and Guoquan Huang

# **Motivation & Contribution**

- GNSS fusion with VIO: Globally accurate and locally precise localization system.
- Utilize raw GNSS measurements simplifying models without losing information.
- Propose a novel way of tightly couple GNSS with VIO by adapting differential GNSS technique to remove atmospheric effects (ionospheric and tropospheric delays).
- Propose 2-step GNSS initialization method that recovers all necessary parameters for raw GNSS fusion.
- Thoroughly evaluated in both simulation and real world data and showed robustness and accuracy of the proposed system.

## IMU, Camera, and raw GNSS Measurements



• IMU: Measures angular velocity and linear acceleration. Used to propagate the state.

$$\mathbf{a}_m = \mathbf{a} + {}^I_W \mathbf{R} \mathbf{g} + \mathbf{b}_a + \mathbf{n}_a, \ \ \boldsymbol{\omega}_m = \boldsymbol{\omega} + \mathbf{c}_m$$

• Camera: Measures feature bearing information. Used to update the state.

$$\mathbf{z}_k = \mathbf{\Pi} ({}_{I}^{C} \mathbf{R}_{W}^{I_k} \mathbf{R} ({}^{W} \mathbf{p}_f - {}^{W} \mathbf{p}_{I_k}) + {}^{C} \mathbf{p}_I$$

- Raw GNSS: Measures dynamics of both GNSS sensor  $\{G\}$  and satellite  $\{S\}$ . Used to update the state.
- Pseudorange

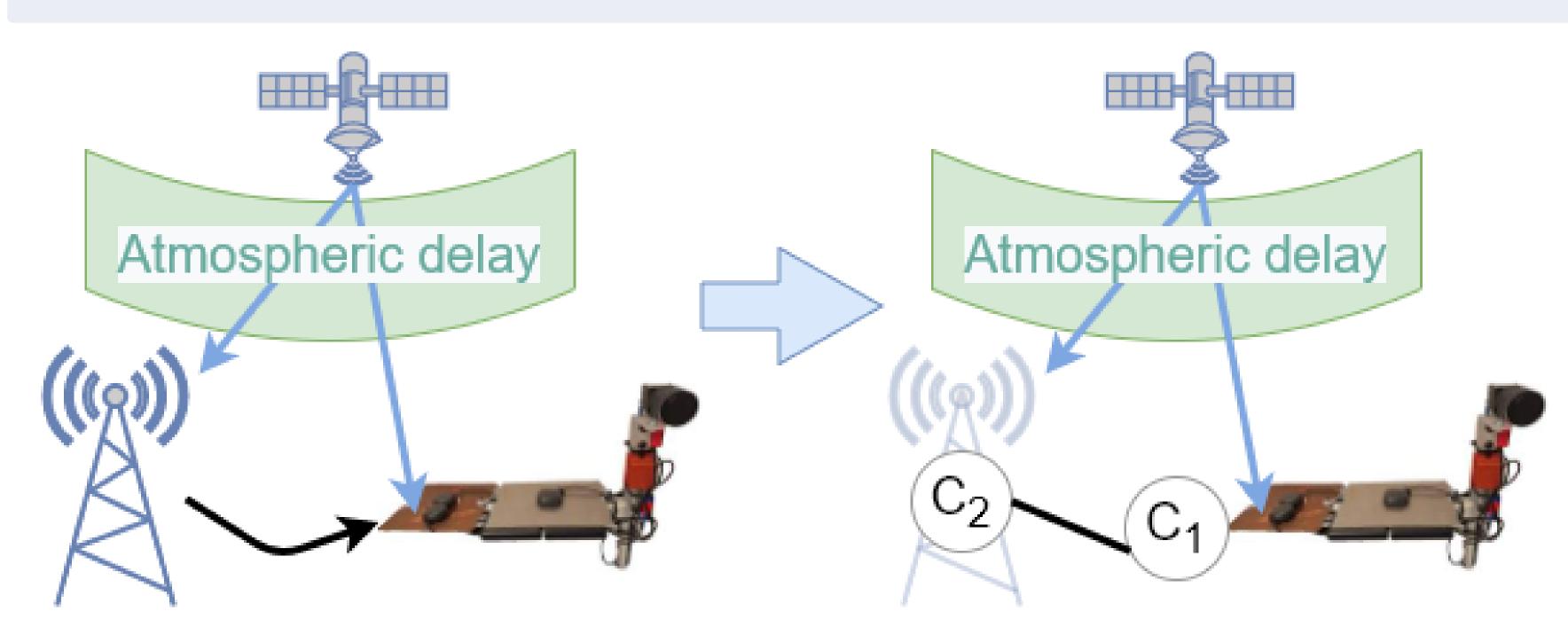
 $z_p = ||^E \mathbf{p}_G - {}^E \mathbf{p}_S||_2 + c(b_G - b_S) + I + T + M + n_p$ – Carrier Phase

 $z_{c} = ||^{E} \mathbf{p}_{G} - {}^{E} \mathbf{p}_{S}||_{2} + c(b_{G} - b_{S}) - I + T + M + \lambda N + n_{c}$ – Doppler Shift

 $z_d = -\left(\left(\mathbf{k}^{\top} ({}^E \mathbf{v}_S - {}^E \mathbf{v}_G) + c(\dot{b}_G - \dot{b}_S)\right)/\lambda + n_d$ Atmospheric delays (I, T) are hard to model.

- $+\mathbf{b}_{g}+\mathbf{n}_{g}$
- $+\mathbf{n}_k$

# **Sequential-Differential GNSS**



- Two sequential GNSS measurements from the same satellite have approximately the same atmospheric delays.
- Subtract two sequential raw GNSS measurements to cancel out atmospheric delays (I, T).
- This is equivalent to performing **differential GNSS** with base station right next to rover.
- Differential Pseudorange

$$D_{p} := z_{p,k+1} - z_{p,k}$$
  
=  $\Delta d + c(b_{G,k+1} - b_{S,k+1}) - c(b_{G,k} - b_{S,k}) + n_{Dp}$ 

• Differential Carrier Phase

$$z_{Dc} := z_{c,k+1} - z_{c,k} = \Delta d + c(b_{G,k+1} - b_{S,k+1}) - c(b_{G,k} - b_{S,k}) + n_{Dc}$$

- Doppler Shift (the same)
- The measurements are now only functions of robot & satellite dynamics and their clock biases.

### **2-Step Initialization**

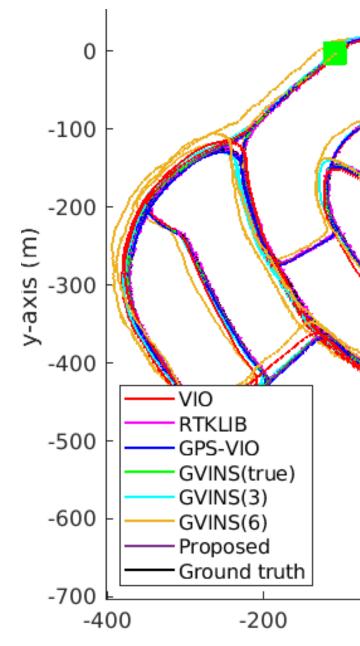
- 0<sup>th</sup>-step: Information collection (GNSS SPP measurements, VIO poses)
- 1<sup>st</sup>-step: ECEF-to-World frame  $\{W\}$  initialization. Find problem
- $2^{nd}$ -step: GNSS sensor parameter (b, b) initialization by solving linear least-squares problem

 $z_d = -\left(\left(\mathbf{k}^{\top} ({}^E \mathbf{v}_S - {}^E \mathbf{v}_G) + c(\dot{b}_G - \dot{b}_S)\right)/\lambda + n_d$ 

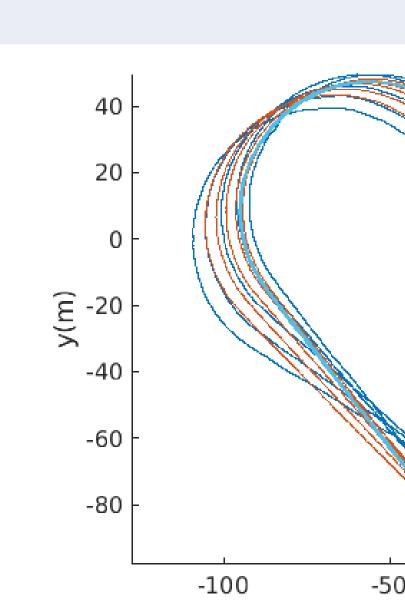
transformation that aligns GNSS and VIO trajectories by solving linear least-squares with quadratic constraint



$dist \setminus \sigma$	<b>0.1m</b>	<b>0.5</b> m	<b>1m</b>	<b>2</b> m	<b>5</b> m
5m	1.57 / 0.58	6.25 / 2.91	14.52/6.79	30.66 / 71.75	69.26 / 88.42
10m	1.31/0.52	5.54/2.19	9.45/4.17	20.41 / 44.94	47.45/94.93
20m	0.79/0.27	2.47 / 0.99	4.84 / 2.01	10.24 / 4.10	26.96 / 51.54
50m	0.53 / 0.07	0.80/0.16	0.97 / 0.27	1.79 / 0.62	4.86 / 1.48
100m	0.45 / 0.09	0.49 / 0.06	0.50/0.12	0.78 / 0.24	2.11/0.65



of atmospheric delays. among all tested



#### • Proposed method showed the smallest RMSE error among all tested

Robotics and Automation (ICRA). IEEE, 2020. on Robotics and Automation (ICRA). IEEE, 2020. IEEE Transactions on Robotics 34.4 (2018): 1004-1020.

**EXAMPLE ICRA 2022 ICRA 2022 IEEE International Conference** 

#### **Simulation Results**

#### • Hyper-parameter sensitivity of initialization (deg/m)

- Evaluated initialization performance with different initialization distances and GNSS SPP noise values.

– Initialization accuracy tend to improve with the longer distance collected and the smaller GNSS SPP noise.

• Localization with different atmospheric delays

	Algorithms	<b>40m RPE</b>
10 20 30 40	OpenVINS [1]	0.08 / 0.26
	GPS-VIO [2]	0.07 / 0.21
	RTKLIB [3]	0.53 / 2.59
	GVINS(true) [4]	0.07 / 0.18
	GVINS(3)	0.46 / 1.21
	GVINS(6)	1.18/3.09
	Proposed	0.076 / 0.185
0 200 400 600 x-axis (m)		

– Proposed method showed **robustness** to different levels

- Proposed method showed the smallest RPE error

#### **Real World Results**

OpenVINS VINS-Mono GVINS RTK GPS-VIO	Algorithms	RMSE(m)
Proposed	VINS-Mono [5]	9.189
	OpenVINS	11.265
	<b>GPS-VIO</b>	0.374
	GVINS	0.327
	Proposed	0.319
0 0 50		

1] Geneva, Patrick, et al. "Openvins: A research platform for visual-inertial estimation." 2020 IEEE International Conference

| Lee, Woosik, et al. "Intermittent gps-aided vio: Online initialization and calibration." 2020 IEEE International Conference [3] T. Takasu and A. Yasuda, "Development of the low-cost rtk-gps receiver with an open source program package rtklib," in

International symposium on GPS/GNSS, vol. 1. International Convention Center Jeju Korea, 2009 [4] Cao, Shaozu, Xiuyuan Lu, and Shaojie Shen. "GVINS: Tightly Coupled GNSS–Visual–Inertial Fusion for Smooth and Consistent State Estimation." IEEE Transactions on Robotics (2022)

[5] Qin, Tong, Peiliang Li, and Shaojie Shen. "Vins-mono: A robust and versatile monocular visual-inertial state estimator."