

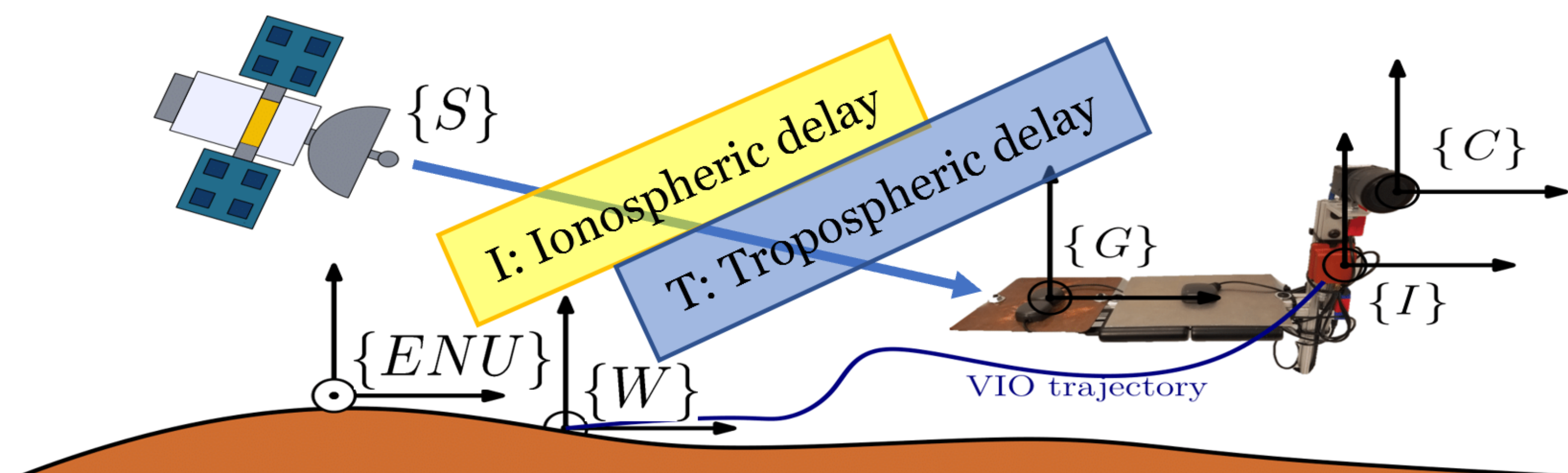
Tightly-coupled GNSS-aided Visual-Inertial Localization

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Motivation & Contribution

- **GNSS fusion with VIO**: Globally accurate and locally precise localization system.
- Utilize **raw GNSS measurements** simplifying models without losing information.
- Propose a novel way of tightly couple GNSS with VIO by adapting differential GNSS technique to **remove atmospheric effects** (ionospheric and tropospheric delays).
- Propose **2-step GNSS initialization** method that recovers all necessary parameters for raw GNSS fusion.
- Thoroughly evaluated in both simulation and real world data and showed robustness and accuracy of the proposed system.

IMU, Camera, and raw GNSS Measurements



- IMU: Measures angular velocity and linear acceleration. Used to propagate the state.

$$\mathbf{a}_m = \mathbf{a} + \mathbf{W}^T \mathbf{R} \mathbf{g} + \mathbf{b}_a + \mathbf{n}_a; \quad \dot{\mathbf{m}} = \dot{\mathbf{I}} + \mathbf{b}_g + \mathbf{n}_g$$

- Camera: Measures feature bearing information. Used to update the state.

$$\mathbf{z}_k = \begin{pmatrix} C \\ I \end{pmatrix} \mathbf{R}_W^{I_k} \mathbf{R} \begin{pmatrix} W \\ W \end{pmatrix} \mathbf{p}_f \quad \mathbf{p}_{I_k} + C \mathbf{p}_I) + \mathbf{n}_k$$

- Raw GNSS: Measures dynamics of both GNSS sensor fGg and satellite fSg . Used to update the state.

- Pseudorange

$$\mathbf{z}_p = \mathbf{j} \mathbf{j}^E \mathbf{p}_G \quad \mathbf{E} \mathbf{p}_{Sj} + c(b_G \quad b_S) + I + T + M + \mathbf{n}_p$$

- Carrier Phase

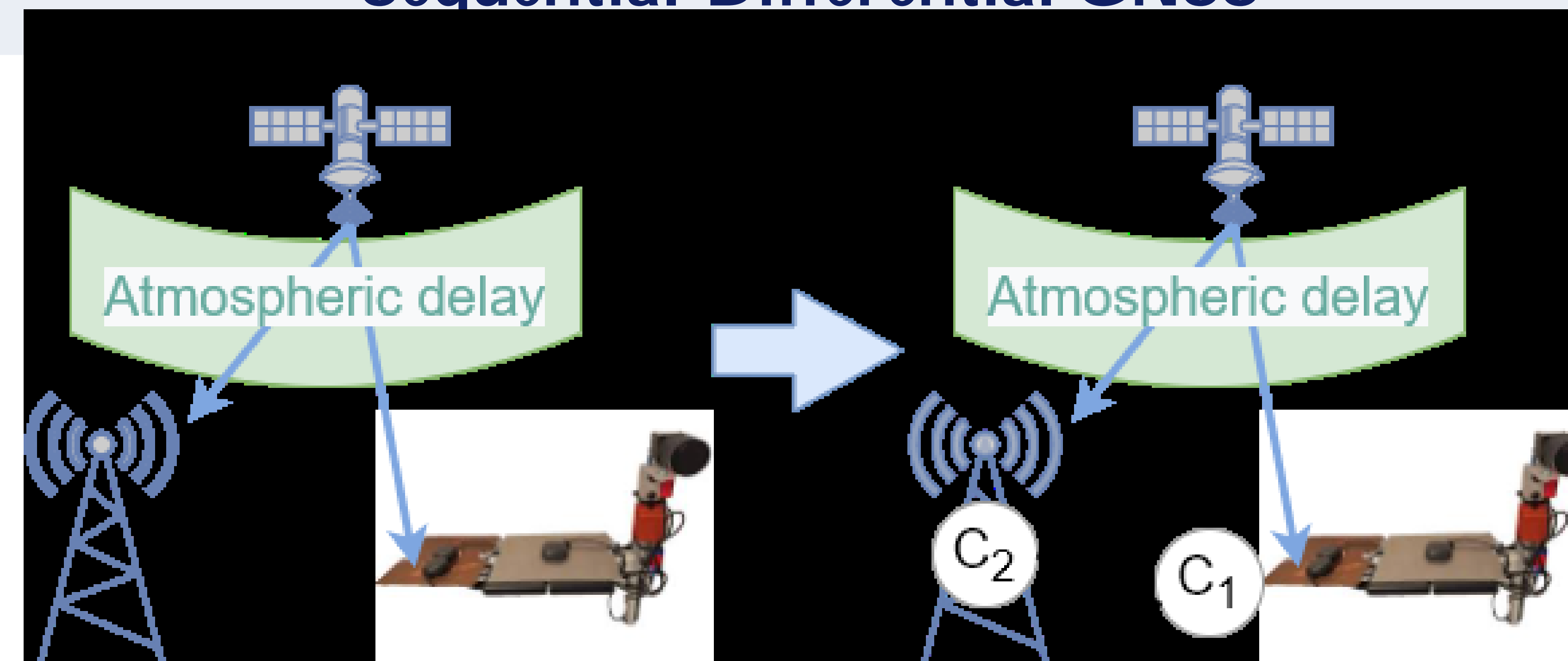
$$\mathbf{z}_c = \mathbf{j} \mathbf{j}^E \mathbf{p}_G \quad \mathbf{E} \mathbf{p}_{Sj} + c(b_G \quad b_S) \quad I + T + M + \quad N + \mathbf{n}_c$$

- Doppler Shift

$$\mathbf{z}_d = (\mathbf{k}^> (\mathbf{E} \mathbf{v}_S \quad \mathbf{E} \mathbf{v}_G) + c(b_G \quad b_S) = + \mathbf{n}_d$$

Atmospheric delays (I; T) are hard to model.

Sequential-Differential GNSS



- Two sequential GNSS measurements from the same satellite have approximately the same atmospheric delays.
- Subtract two **sequential** raw GNSS measurements to **cancel out atmospheric delays (I, T)**.
- This is equivalent to performing **differential GNSS** with base station right next to rover.

- Differential Pseudorange

$$\begin{aligned} Z_{Dp} &:= Z_{p;k+1} \quad Z_{p;k} \\ &= d + c(b_{G;k+1} \quad b_{S;k+1}) \quad c(b_{G;k} \quad b_{S;k}) + n_{Dp} \end{aligned}$$

- Differential Carrier Phase

$$\begin{aligned} Z_{Dc} &:= Z_{c;k+1} \quad Z_{c;k} \\ &= d + c(b_{G;k+1} \quad b_{S;k+1}) \quad c(b_{G;k} \quad b_{S;k}) + n_{Dc} \end{aligned}$$

- Doppler Shift (the same)

$$\mathbf{z}_d = (\mathbf{k}^> (\mathbf{E} \mathbf{v}_S \quad \mathbf{E} \mathbf{v}_G) + c(b_G \quad b_S) = + \mathbf{n}_d$$

- The measurements are now only functions of **robot & satellite dynamics and their clock biases**.

2-Step Initialization

- 0th-step: Information collection (GNSS SPP measurements, VIO poses)
- 1st-step: ECEF-to-World frame fWg initialization. Find transformation that aligns GNSS and VIO trajectories by solving linear least-squares with quadratic constraint problem
- 2nd-step: GNSS sensor parameter ($b; \dot{b}$) initialization by solving linear least-squares problem

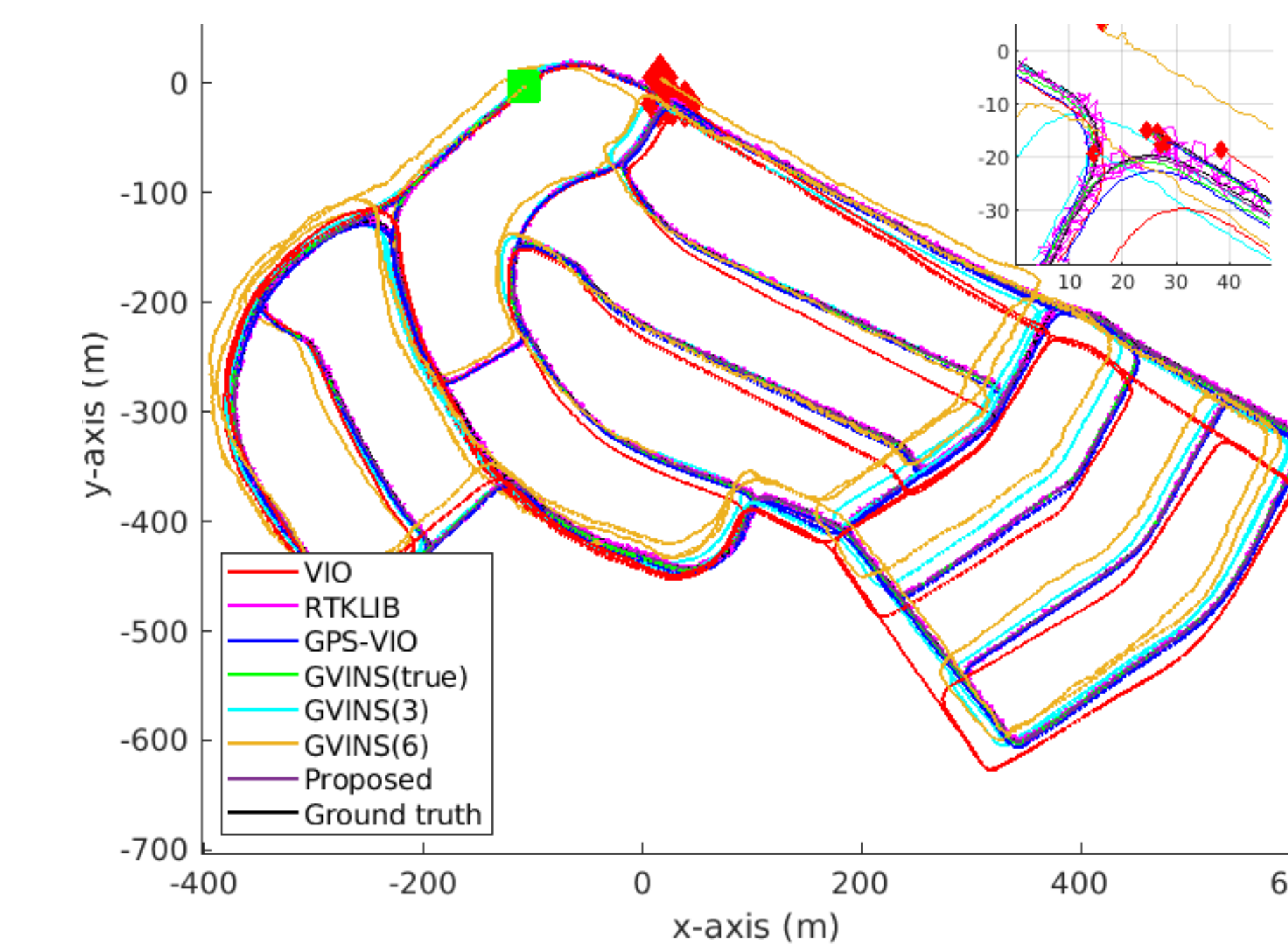
Simulation Results

- Hyper-parameter sensitivity of initialization (deg/m)

dist/n	0.1m	0.5m	1m	2m	5m
5m	1.57 / 0.58	6.25 / 2.91	14.52 / 6.79	30.66 / 71.75	69.26 / 88.42
10m	1.31 / 0.52	5.54 / 2.19	9.45 / 4.17	20.41 / 44.94	47.45 / 94.93
20m	0.79 / 0.27	2.47 / 0.99	4.84 / 2.01	10.24 / 4.10	26.96 / 51.54
50m	0.53 / 0.07	0.80 / 0.16	0.97 / 0.27	1.79 / 0.62	4.86 / 1.48
100m	0.45 / 0.09	0.49 / 0.06	0.50 / 0.12	0.78 / 0.24	2.11 / 0.65

- Evaluated initialization performance with different initialization distances and GNSS SPP noise values.
- Initialization accuracy tend to improve with **the longer distance** collected and **the smaller GNSS SPP noise**.

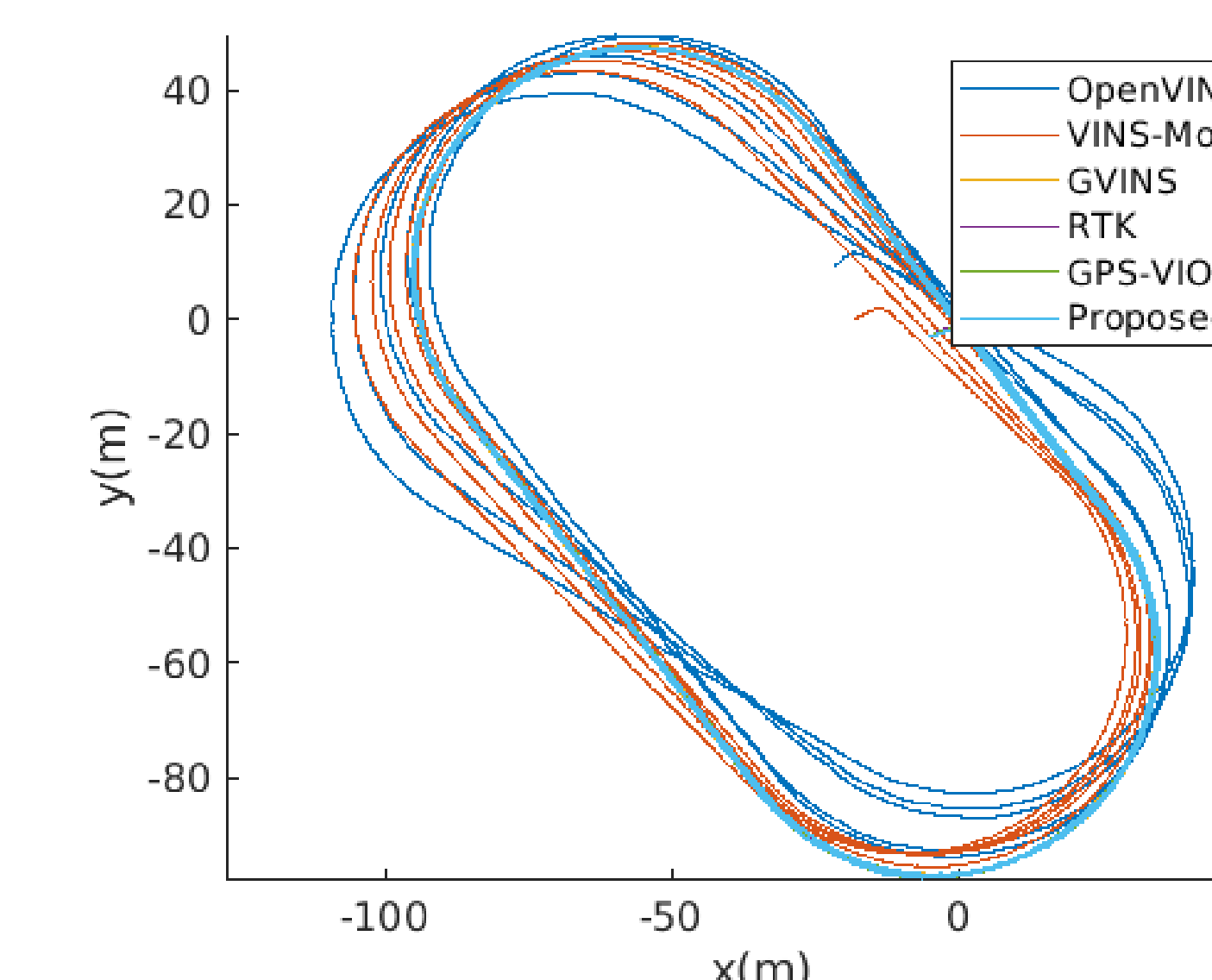
- Localization with different atmospheric delays



Algorithms	40m RPE
OpenVINS [1]	0.08 / 0.26
GPS-VIO [2]	0.07 / 0.21
RTKLIB [3]	0.53 / 2.59
GVINS(true) [4]	0.07 / 0.18
GVINS(3)	0.46 / 1.21
GVINS(6)	1.18 / 3.09
Proposed	0.076 / 0.185

- Proposed method showed **robustness** to different levels of atmospheric delays.
- Proposed method showed **the smallest RPE error** among all tested

Real World Results



Algorithms	RMSE(m)
VINS-Mono [5]	9.189
OpenVINS	11.265
GPS-VIO	0.374
GVINS	0.327
Proposed	0.319

- Proposed method showed **the smallest RMSE error** among all tested

[1] Geneva, Patrick, et al. "OpenVINS: A research platform for visual-inertial estimation." 2020 IEEE International Conference on Robotics and Automation (ICRA). IEEE, 2020.
 [2] Lee, Woosik, et al. "Intermittent gps-aided vio: Online initialization and calibration." 2020 IEEE International Conference on Robotics and Automation (ICRA). IEEE, 2020.
 [3] T. Takasu and A. Yasuda, "Development of the low-cost rtk-gps receiver with an open source program package rtklib," in International symposium on GPS/GNSS, vol. 1. International Convention Center Jeju Korea, 2009
 [4] Cao, Shaozu, Xiuyuan Lu, and Shaojie Shen. "GVINS: Tightly Coupled GNSS-Visual-Inertial Fusion for Smooth and Consistent State Estimation." IEEE Transactions on Robotics (2022).
 [5] Qin, Tong, Peiliang Li, and Shaojie Shen. "Vins-mono: A robust and versatile monocular visual-inertial state estimator." IEEE Transactions on Robotics 34.4 (2018): 1004-1020.